



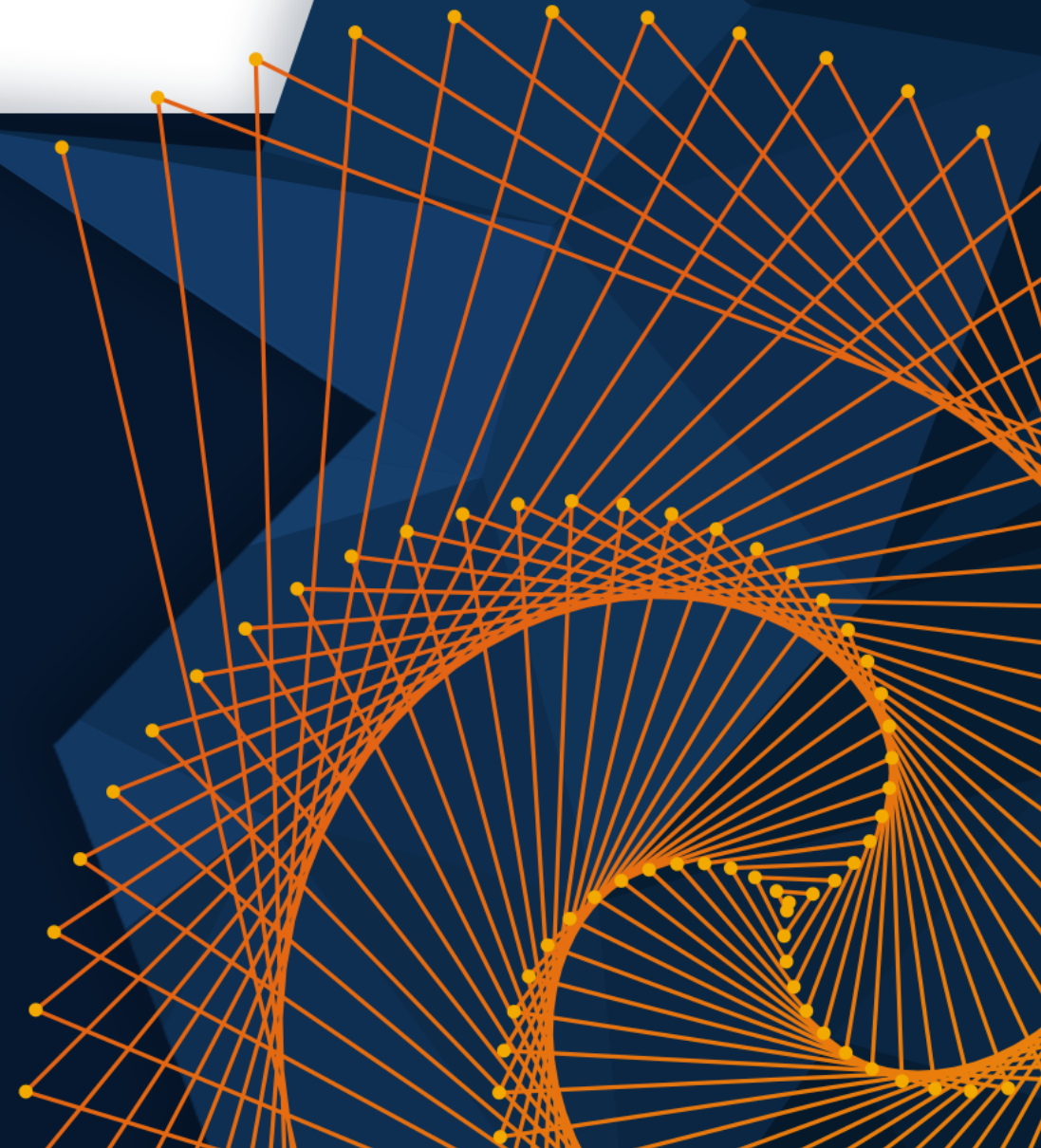
November 13–14, 2024 | Online

Spacecraft Power Capability (SPoC) Model for NASA Artemis Missions

Hector Hernandez, NASA

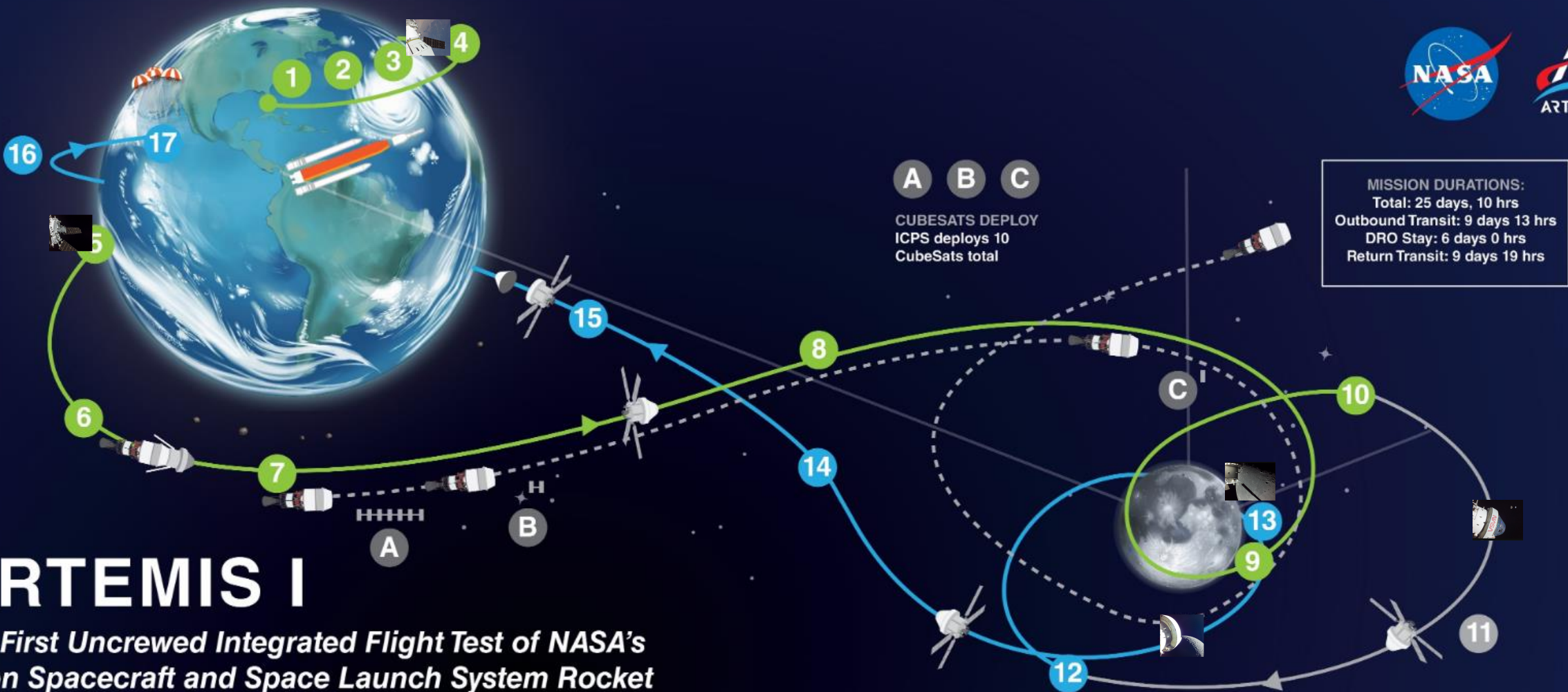


MATLAB EXPO



Agenda

- Artemis 1 Mission Overview
- Spacecraft Power Capability (SPoC) Background
- Example Challenges
 - Solar Array Performance Modeling
 - Battery Model Updates
 - Mission Planning & Design
- Conclusion



ARTEMIS I

The First Uncrewed Integrated Flight Test of NASA's Orion Spacecraft and Space Launch System Rocket

- 1 LAUNCH (11/16/22)**
SLS and Orion lift off from pad 39B at Kennedy Space Center.
- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM**
- 3 CORE STAGE MAIN ENGINE CUT OFF**
With separation.
- 4 PERIGEE RAISE MANEUVER**
- 5 EARTH ORBIT**
Systems check with solar panel adjustments.
- 6 TRANS LUNAR INJECTION (TLI) BURN**
Maneuver lasts for approximately 20 minutes.
- 7 INTERIM CRYOGENIC PROPULSION STAGE (ICPS) SEPARATION AND DISPOSAL**
ICPS commits Orion to moon at TLI.
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Pacific Ocean landing within view of the U.S. Navy recovery ship.

Flight Day 1 Post Solar Array Deployment

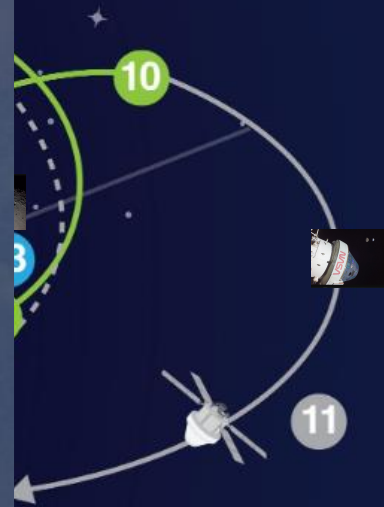
ARTEMIS

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- 6 TRANS LUNAR INJECTION Maneuver approximately 100,000 miles from Earth.

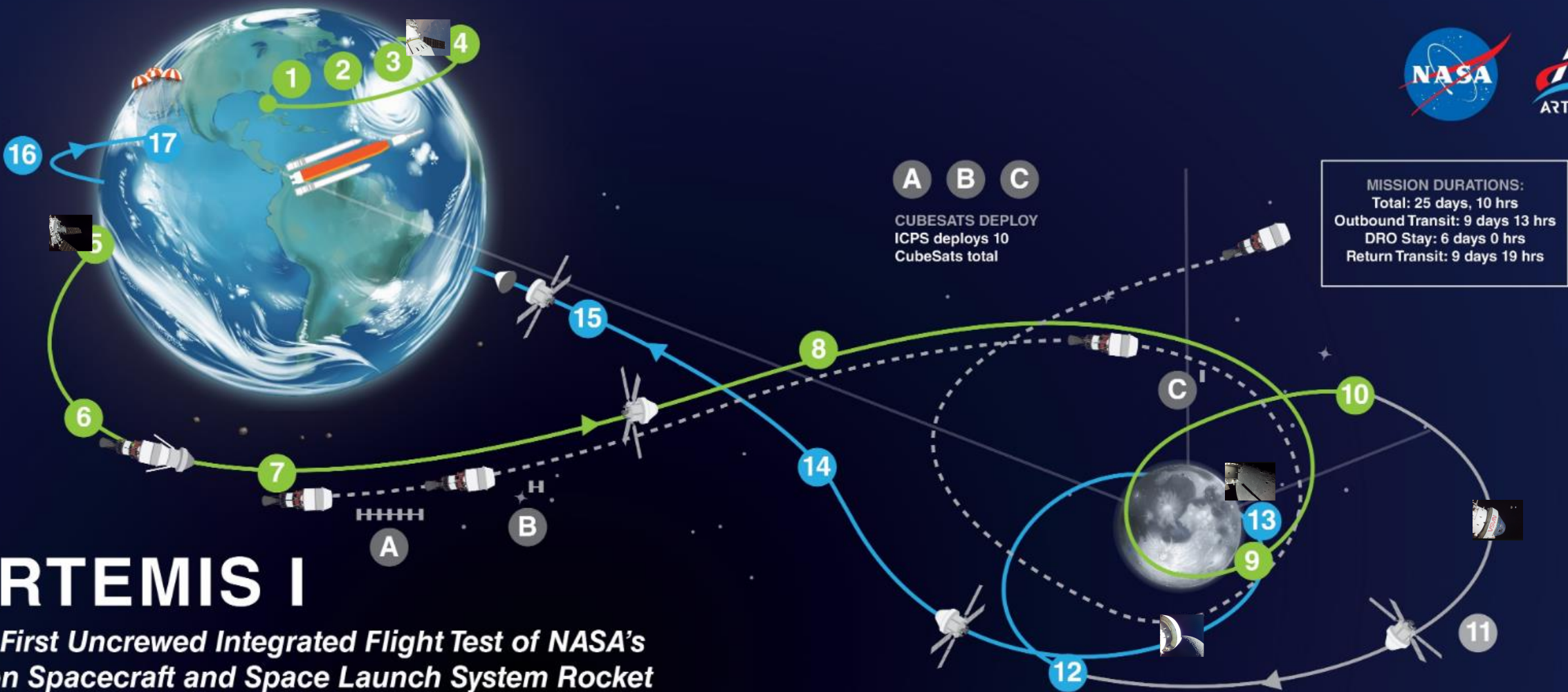


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retrograde orbit (DRO). (9 day duration) in the orbit 45,700 miles from the surface of the Moon. Burns as necessary to aim for Earth's atmosphere.



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4 Flight Day 1 Service Module Inspection

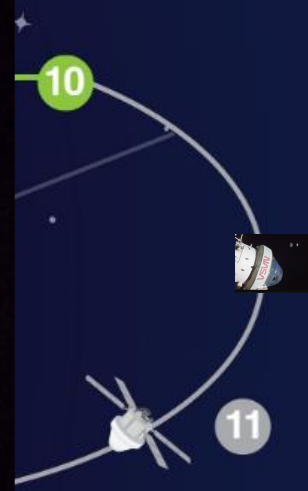
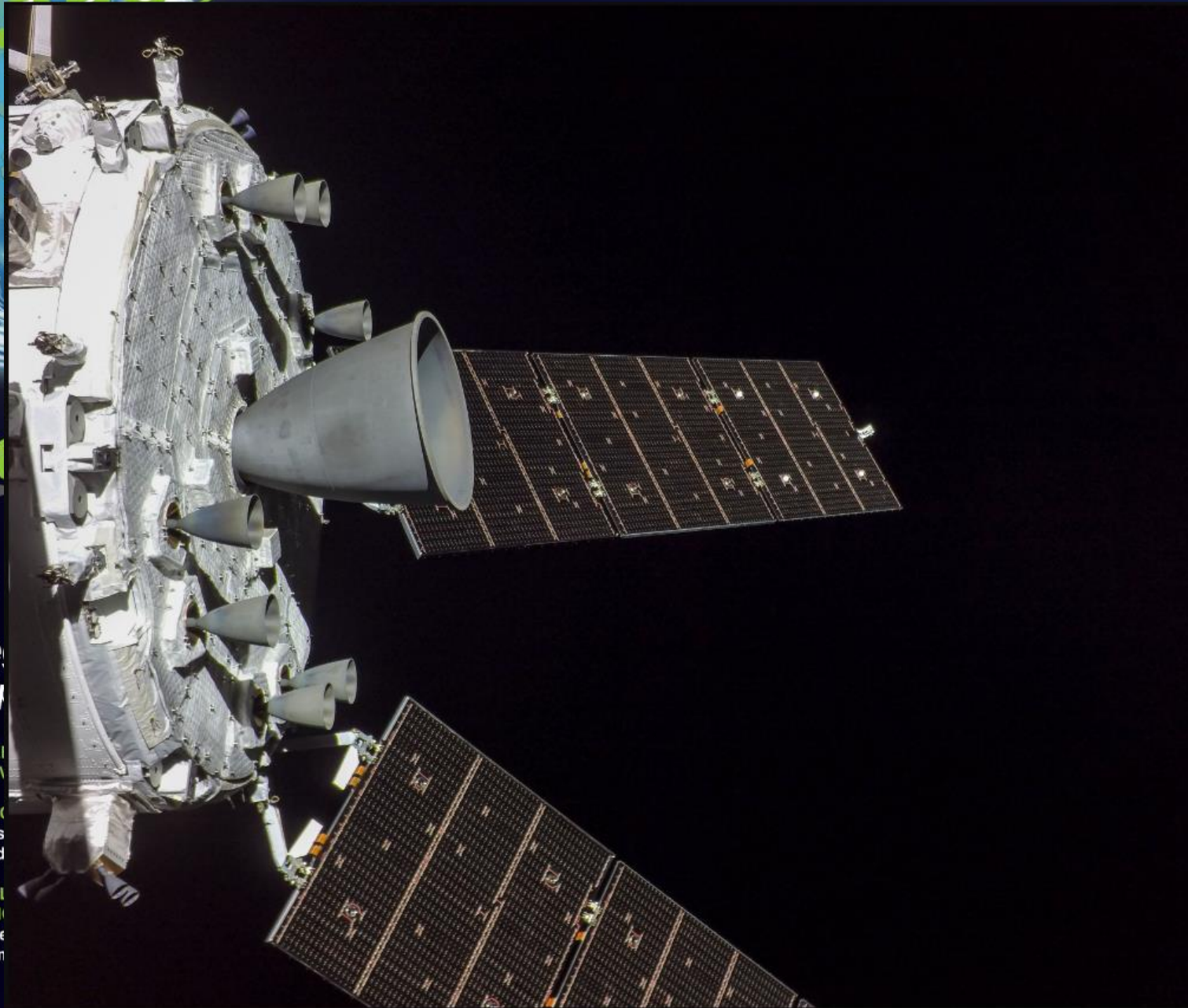


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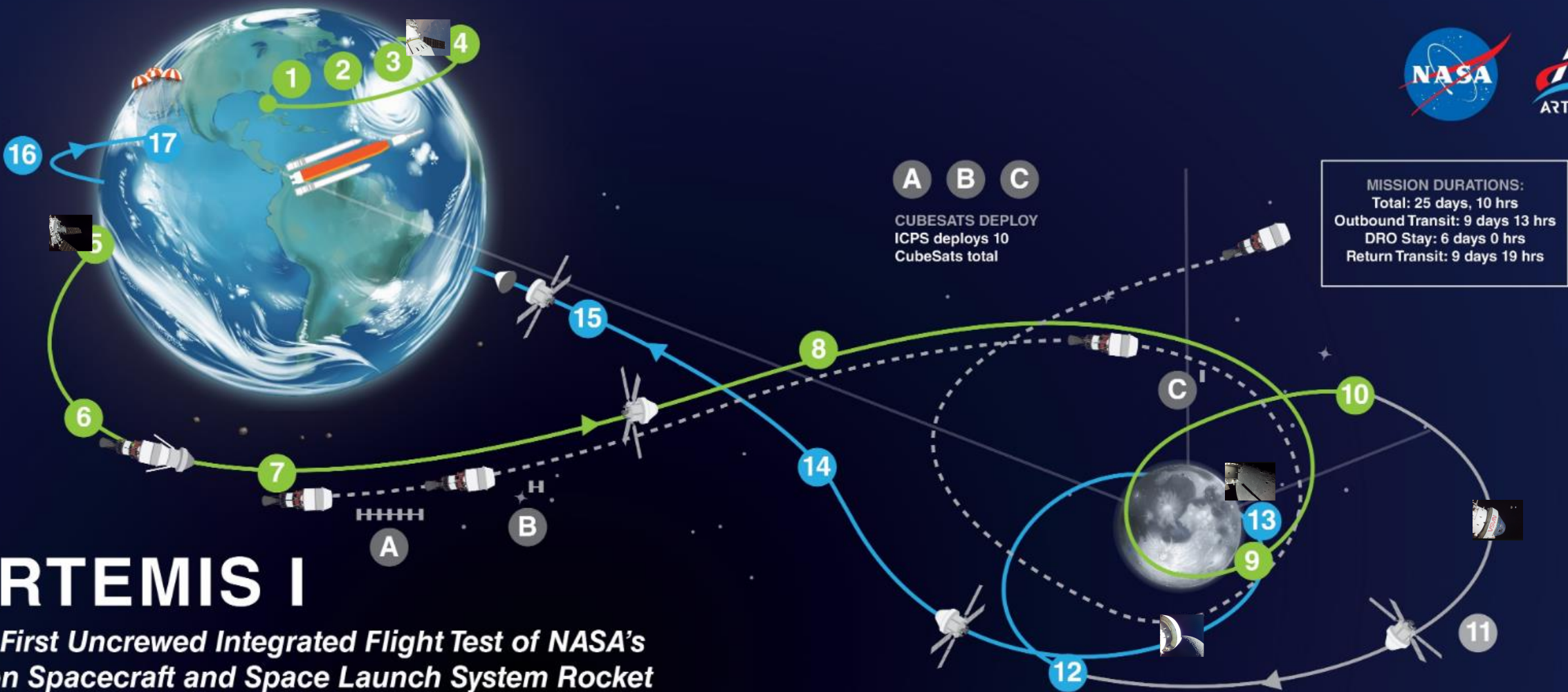
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- 6 TRANS INJECTION MANEUVER
Maneuver to approximate lunar orbit.



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Flight Day 13 Orion Selfie



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ARTEMIS

The First Uncrewed International
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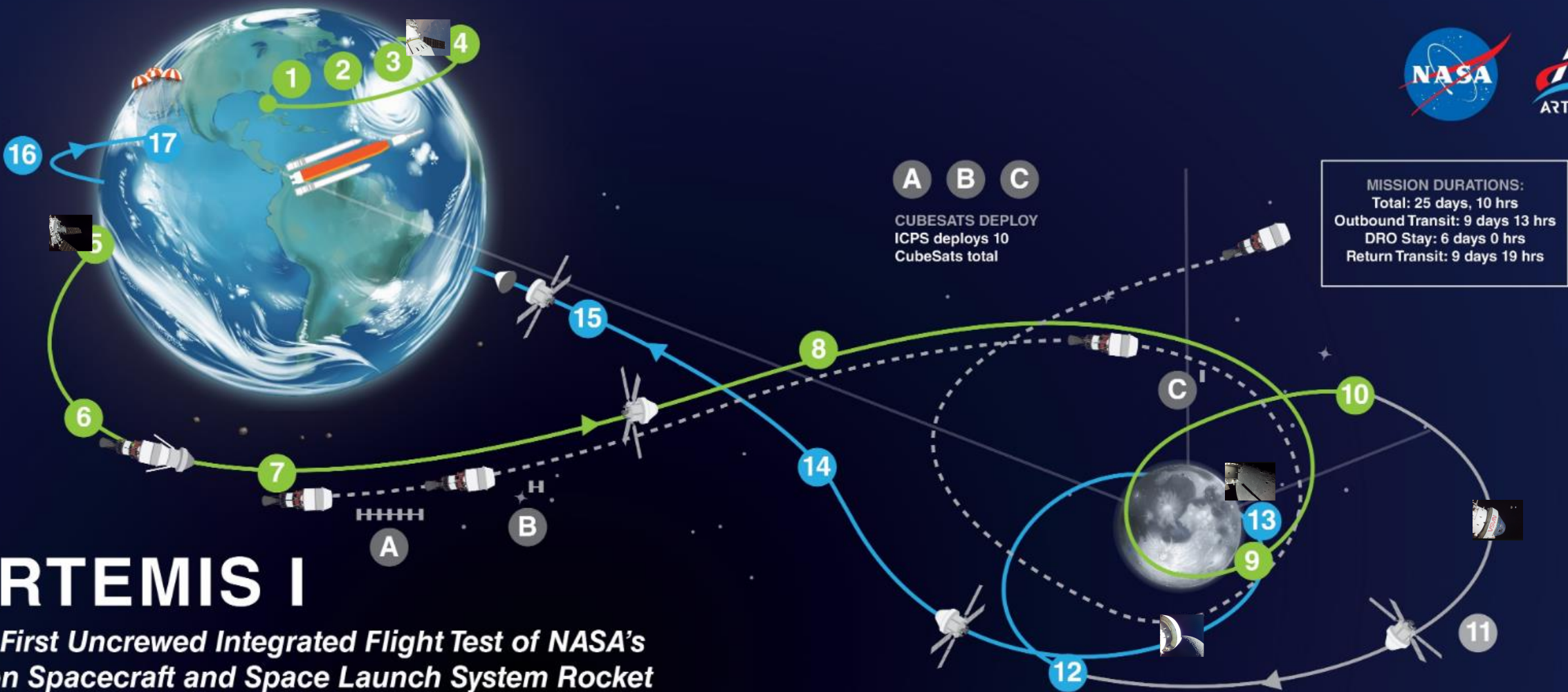
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- 5 EARTH ORBIT System panel activation
- 6 TRANSITION INJECT MANEUVER
Maneuver approximately 100 miles from Earth.



CREW MODULE SEPARATION FROM SERVICE MODULE

ENTRY INTERFACE
Enter Earth's atmosphere.

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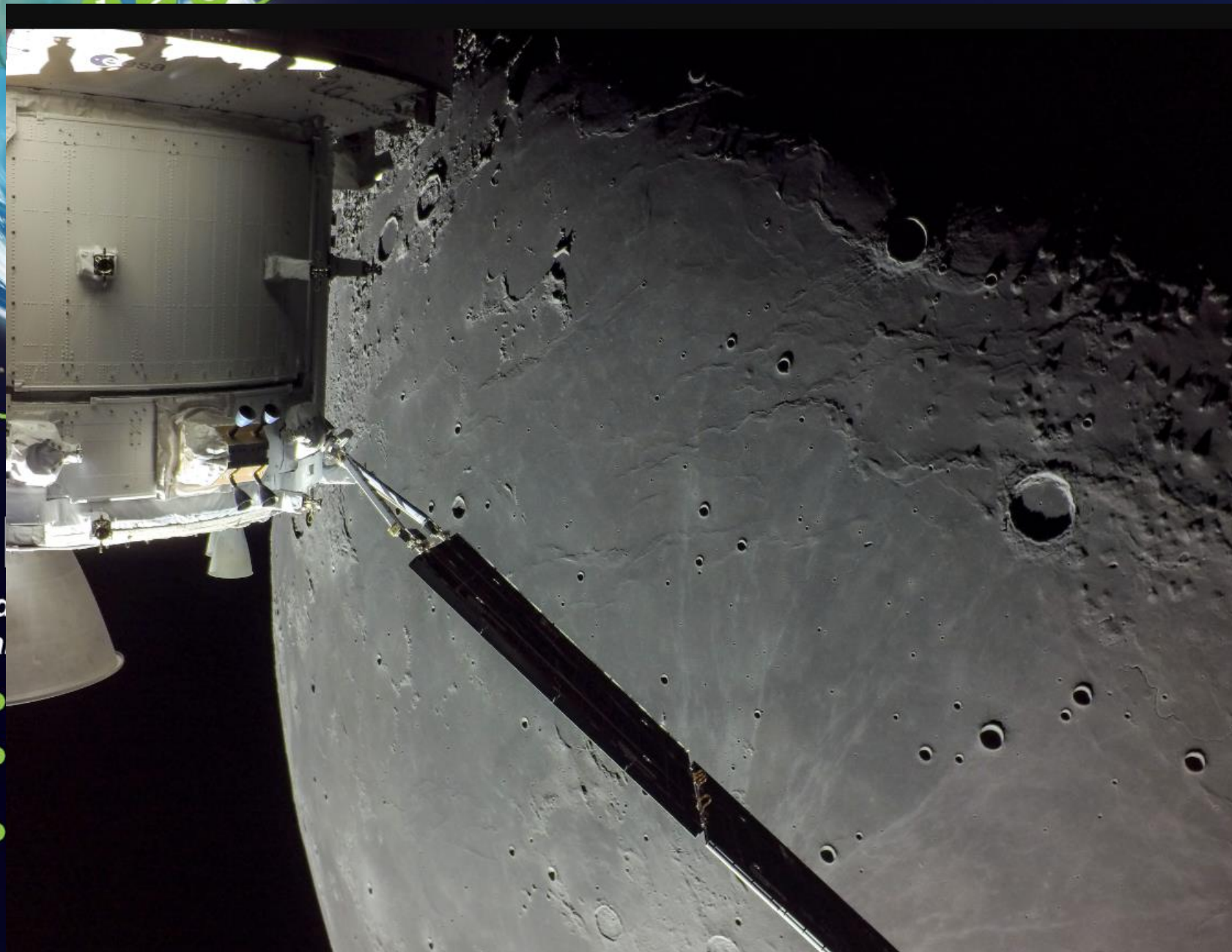
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Flight Day 20 Pre Return Powered Flyby

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 LRO Stay: 6 days 0 hrs
 Lunar Transit: 9 days 19 hrs



16

17

5

6

ARTEMIS

The First Uncrewed Orion Spacecraft

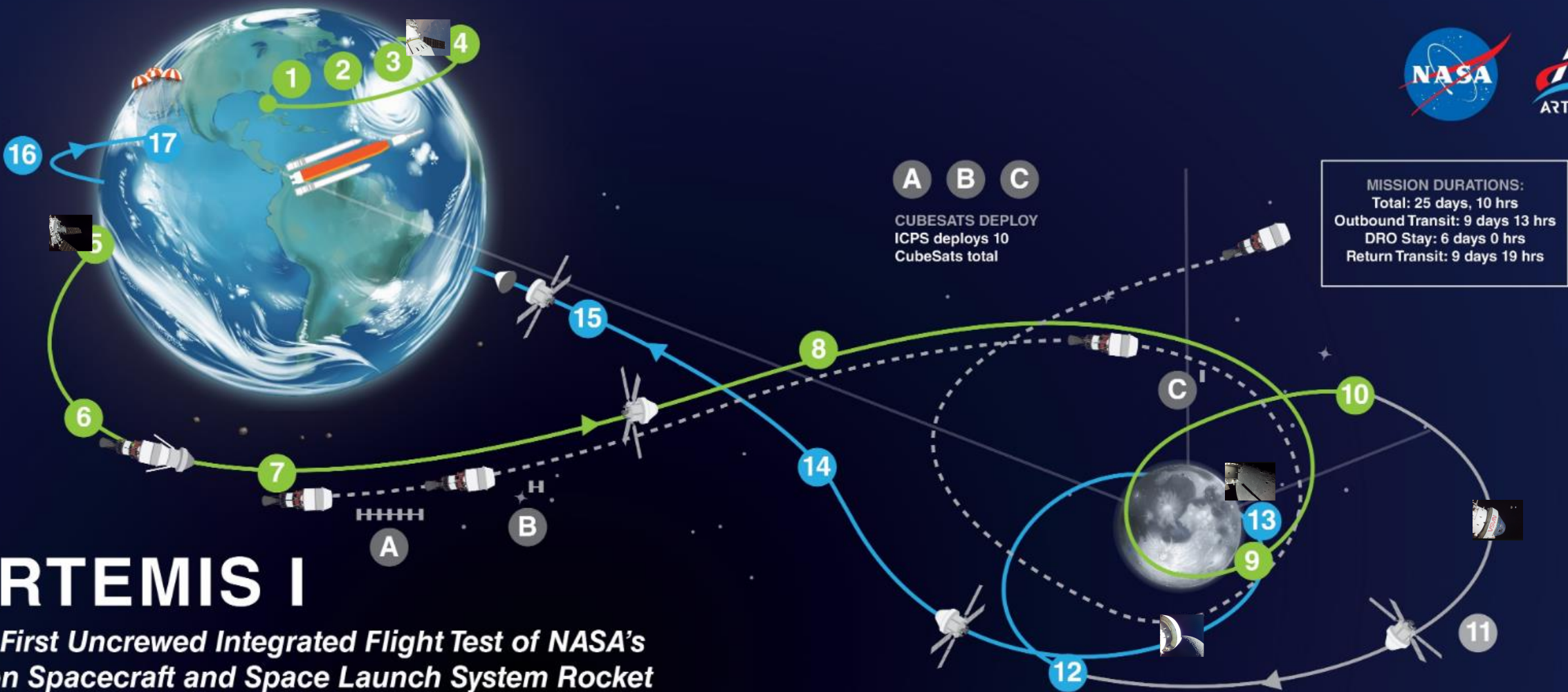
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Flight Day 20 Post Return Powered Flyby

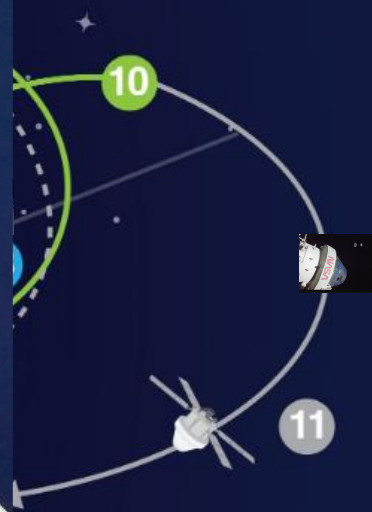
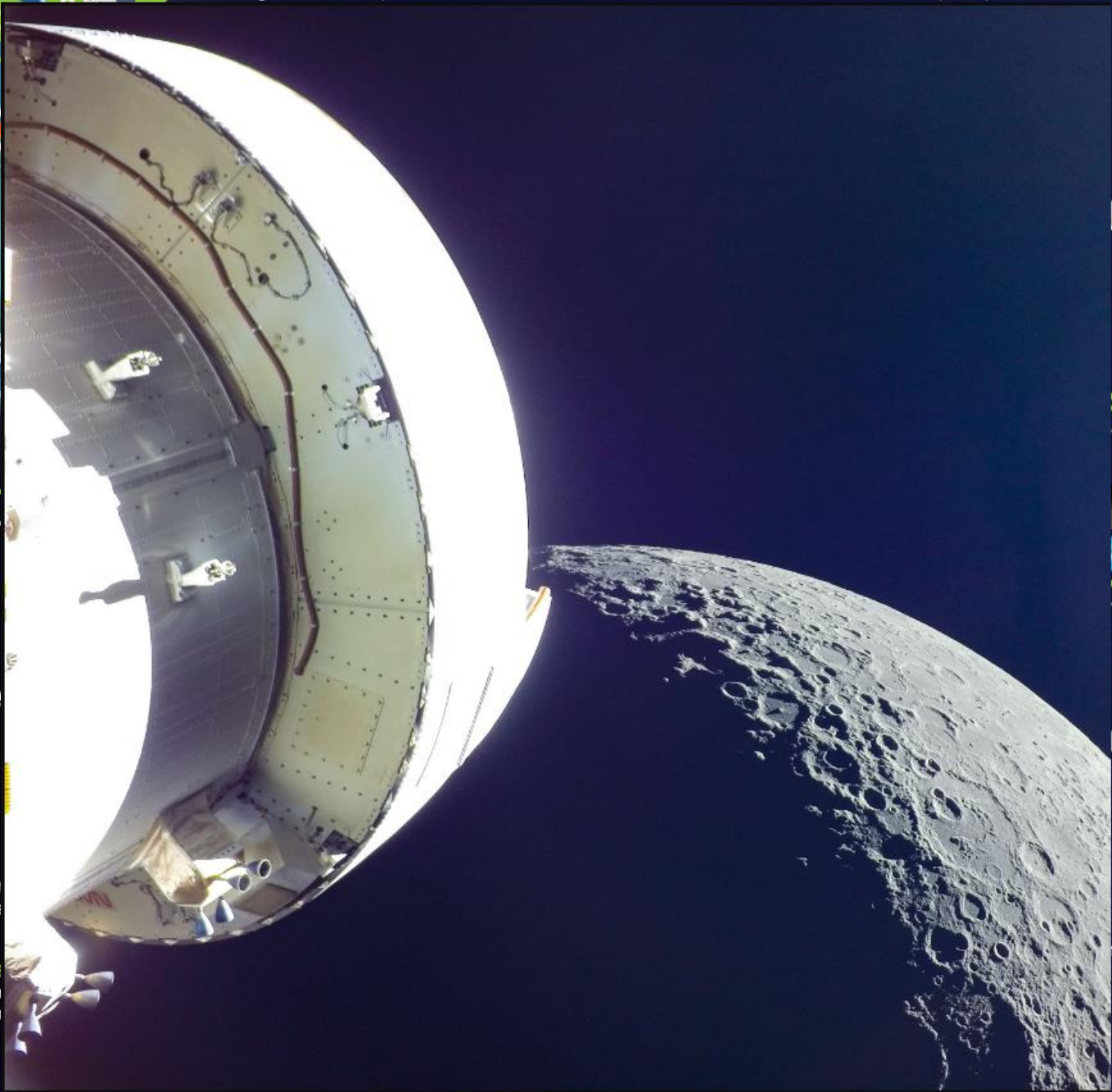


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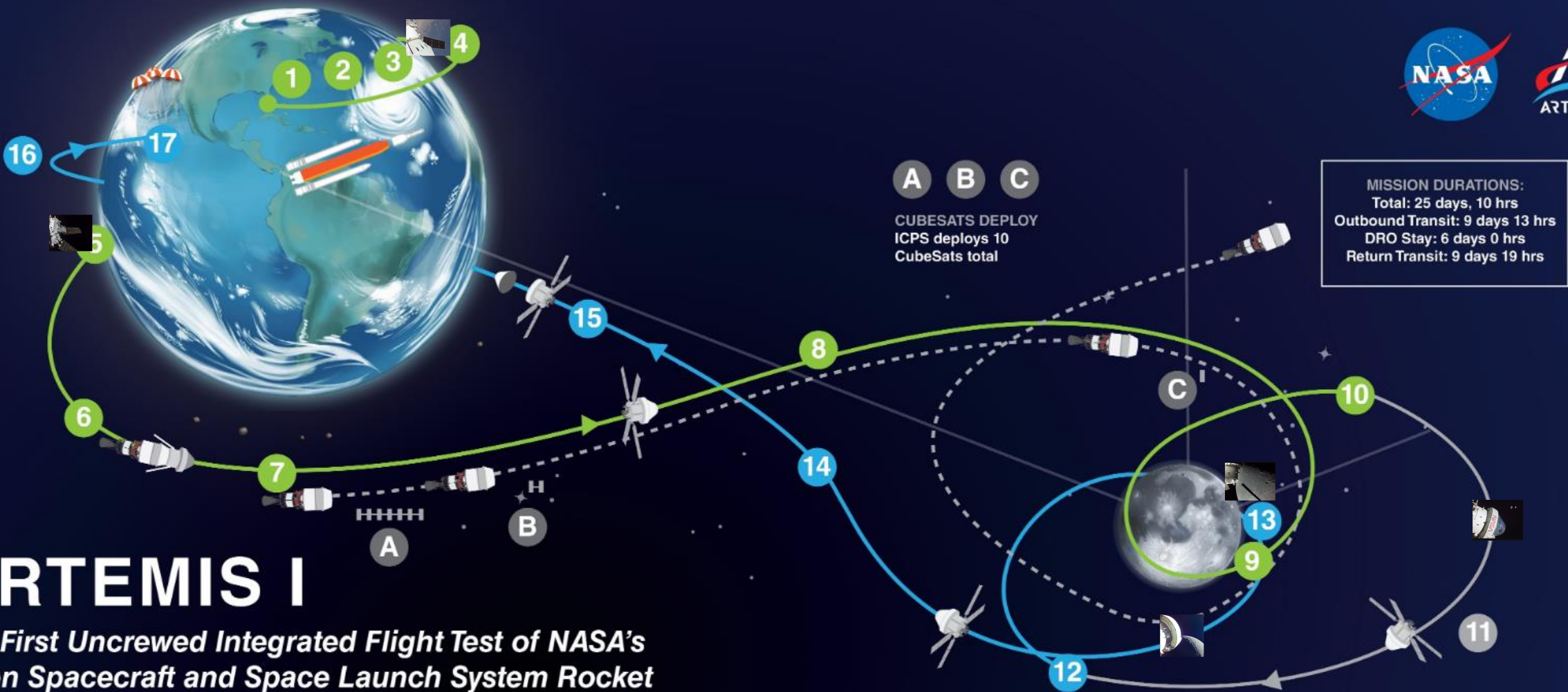
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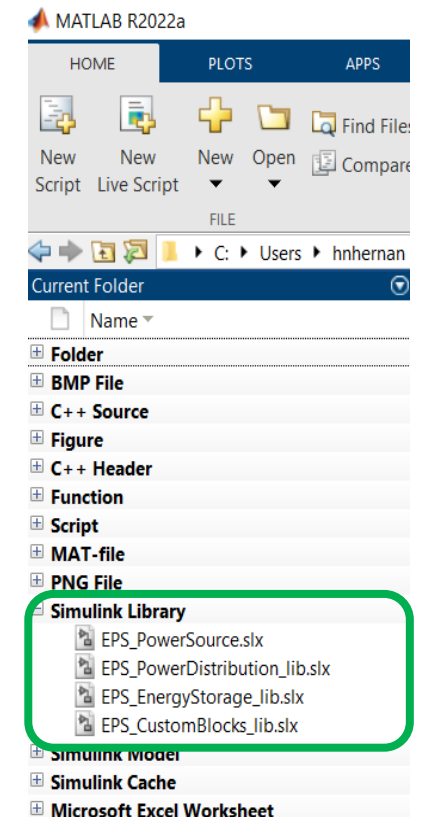
SPoC Background

- **Spacecraft Power Capability (SPoC) is a computer model originally developed for predicting the electrical performance of the Orion Exploration Flight Test-1 (EFT-1) Electrical Power Subsystem (EPS)**
 - Developed in Simulink® environment using Simscape™
 - Mix of validation techniques used in a complementary fashion to reduce risk of errors
- **Integrated power analysis (IPA) to support mission level design and vehicle performance analysis**
 - Supported the Orion Artemis 1 prelaunch, real-time and post-flight analysis
 - Currently supports Orion Artemis 2+ and other NASA programs
- **Verification and validation of EPS requirements and flight test objectives**
- **SPoC determines EPS capability and predicts the system state**
 - Power generation, battery state of charge (SoC), voltage, etc.



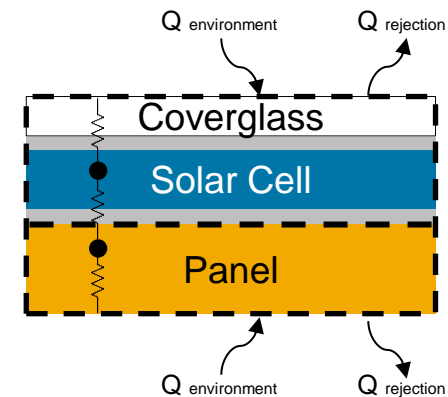
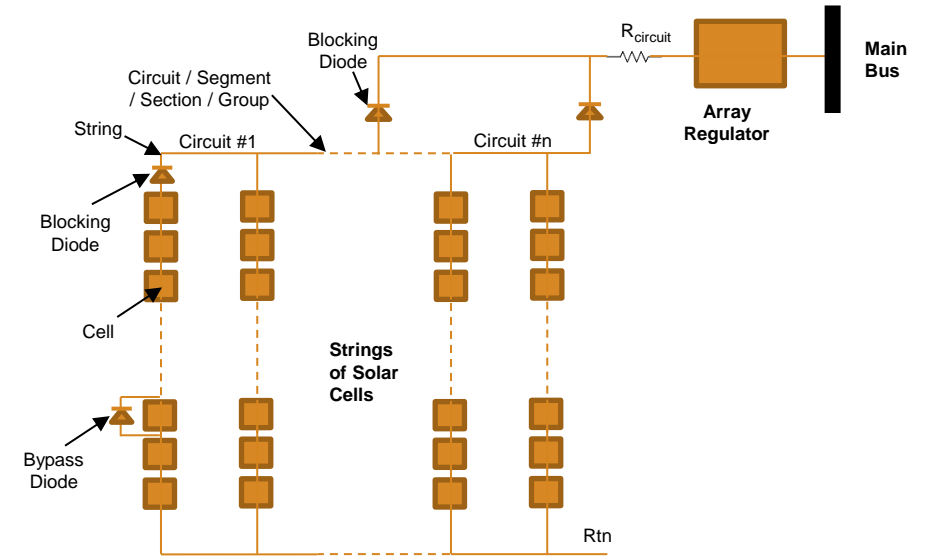
SPoC Background Cont'd

- **SPoC has three major component libraries for spacecraft power system applications**
 - *Power Source/Generation library*
 - Orion heritage-based solar cell model similar to the standard Hughes Model
 - Customized for single cell, string or segment operation and robust modeling convergence
 - *Energy Storage Library*
 - Orion heritage-based equivalent circuit models of a cell or battery
 - *Power Management & Distribution Library*
 - Generic power distribution unit templates that includes power inputs and outputs
Usually, the most customized component for representing different power system architectures
 - Heritage-based cabling models, fuses, switches, regulators
- **Major components are built using custom blocks from the *EPS Custom Blocks Library* and have been tested & validated**
 - Custom blocks are coded using Simscape™ Language

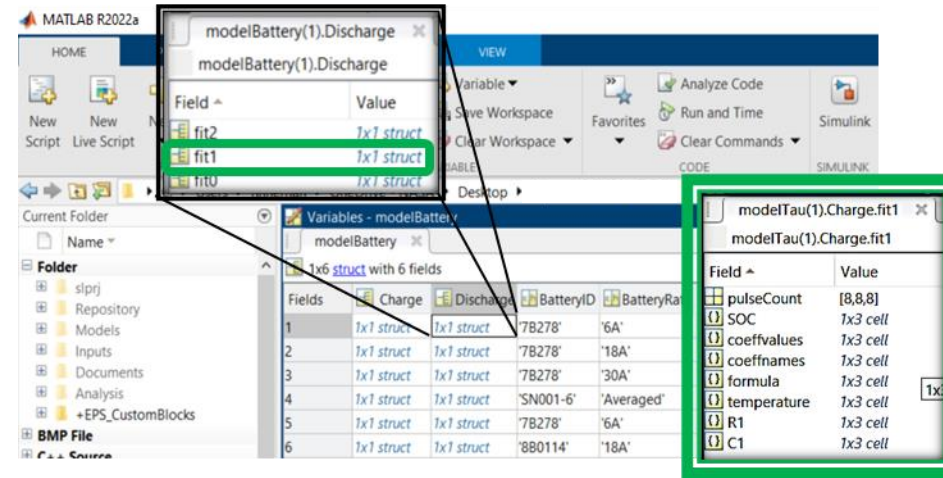
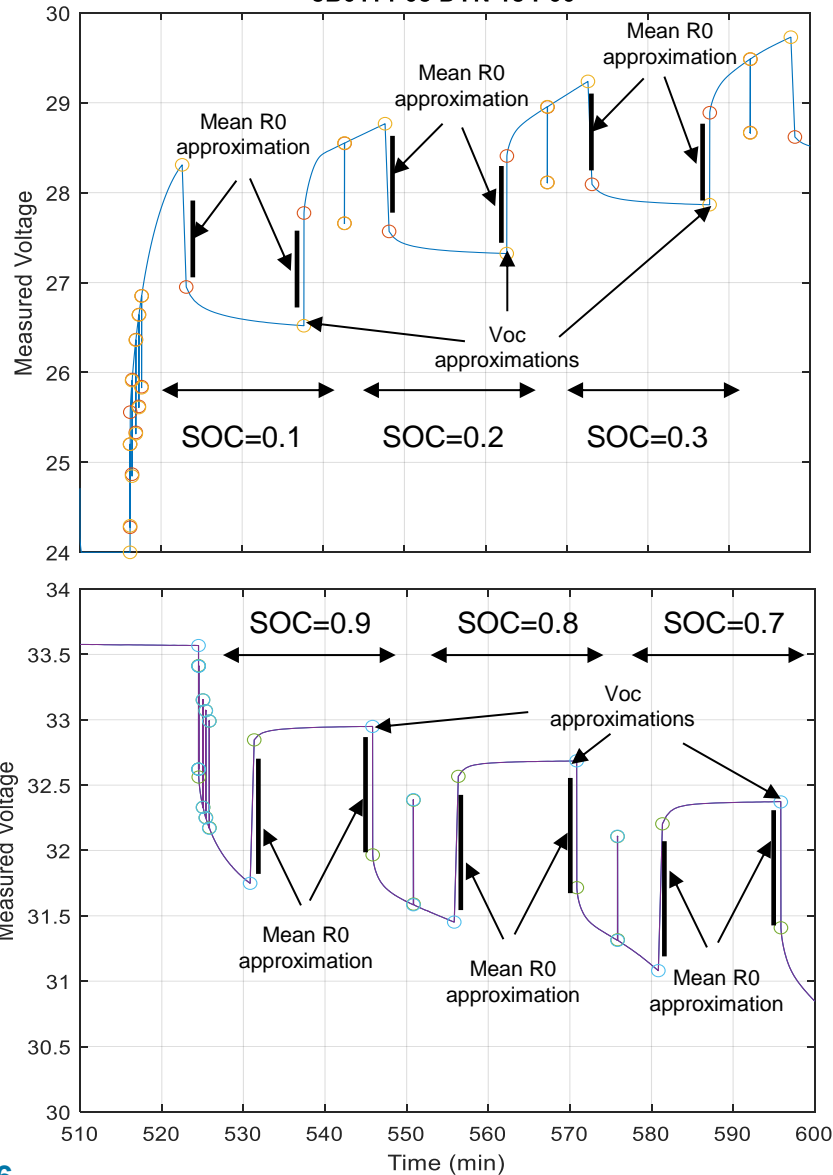


Challenges: Solar Array Performance Modeling

- **Modeling solar array performance requires knowledge of mission and vehicle level conditions such as:**
 - Operating environment (LEO, GEO, LLO, etc.) & duration
 - Vehicle orientation (off-pointed conditions, off-nominal attitudes)
 - Solar flux and thermal conditions
 - Cell/wing level electrical and thermal parameters
- **SPOC models the entire solar array electrical design**
 - Discrete components from cells up to the array regulator
 - Strings connected to proper channels based on drawings
- **SPOC models the solar array operating temperatures based on mission & vehicle conditions**
 - Simplified thermal model that can accommodate different types of solar cells & substrates (flex blankets & rigid panels)
 - Time-phased predicts of array temperature in eclipse & sunlight



Challenges: Battery Model Updates



- **Processing test data for battery model parameter estimation can be challenging and time-consuming**
 - Depends on the size and quality of the data
- **Different applications require different degrees of battery model fidelity**
 - Mission level analysis (steady state voltage drop, energy balance)
 - System level analysis (power quality)
 - Dedicated State of Charge (SOC) and State of Health (SOH) estimation
- **Consequently, the same dataset(s) are processed multiple different times until the next update cycle is scheduled to occur**
 - Highly inefficient and works against completing scheduled milestones
- **The solution: auto-processed dataset(s) in MATLAB for all applications at one time until the next update cycle occurs or is required**

Challenges: Power Model Validation

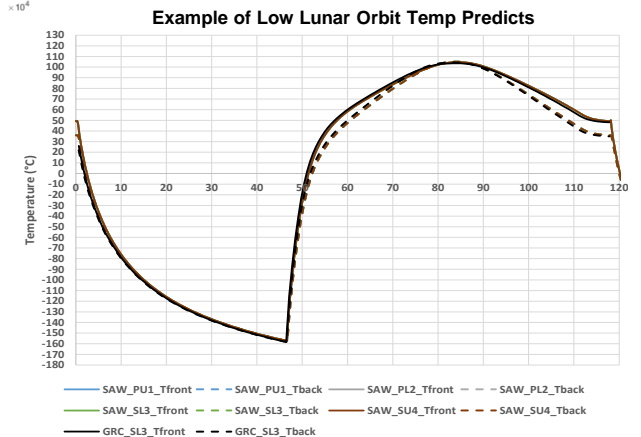
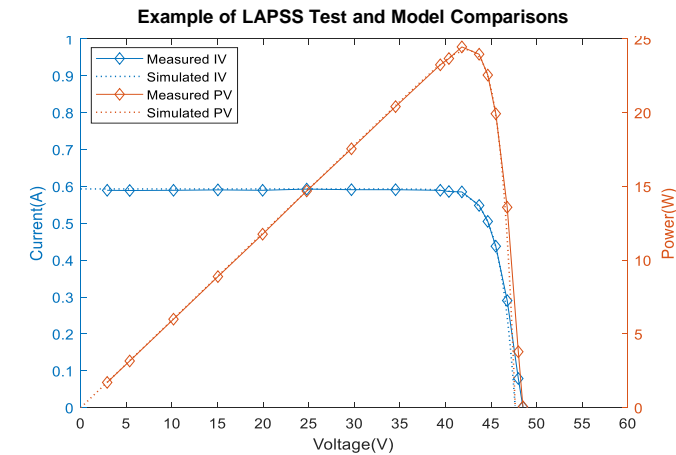
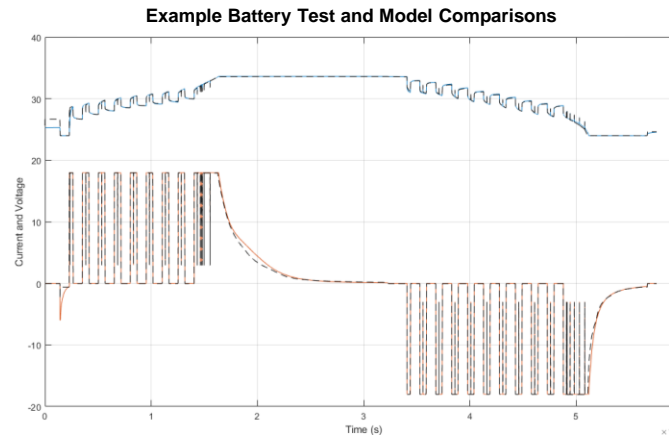
- **Model validation process and activities to show objective evidence that the model reflects the system as accurately as necessary**

- **Implementation:**

- Inspection of reasonable outputs
- Internal peer reviews of model underpinnings
- Measured data and simulation comparisons

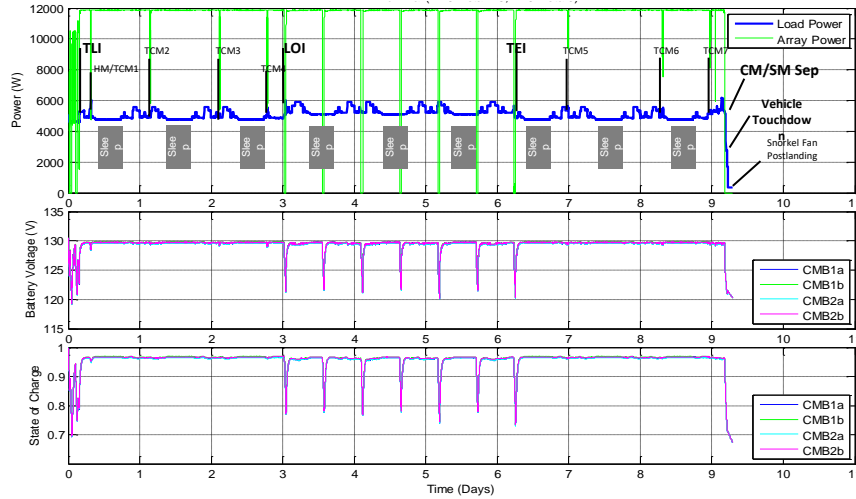
- **Outcomes:**

- The model behaves as expected with negligible errors
- Some modifications were required followed by revalidation to confirm corrections
- Some differences between the model and the real system which cannot be corrected but are quantified and understood



Challenges: Mission Design & System Performance

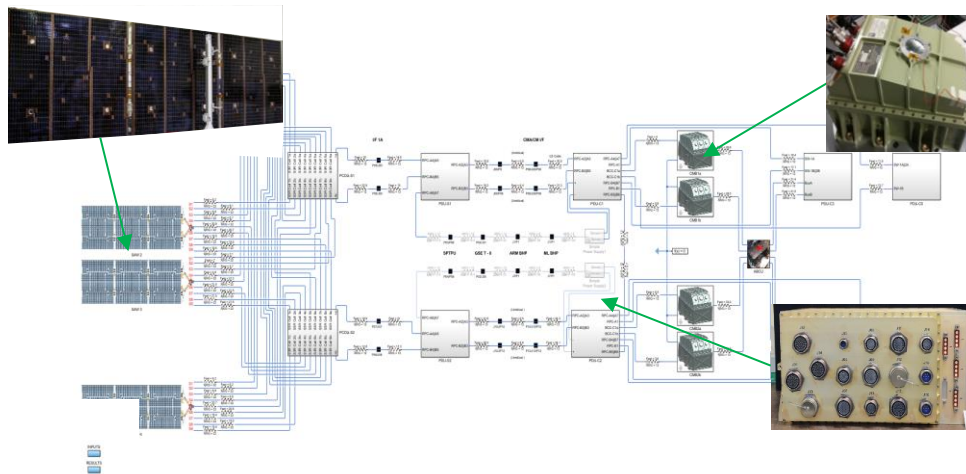
Integrated Power Analysis



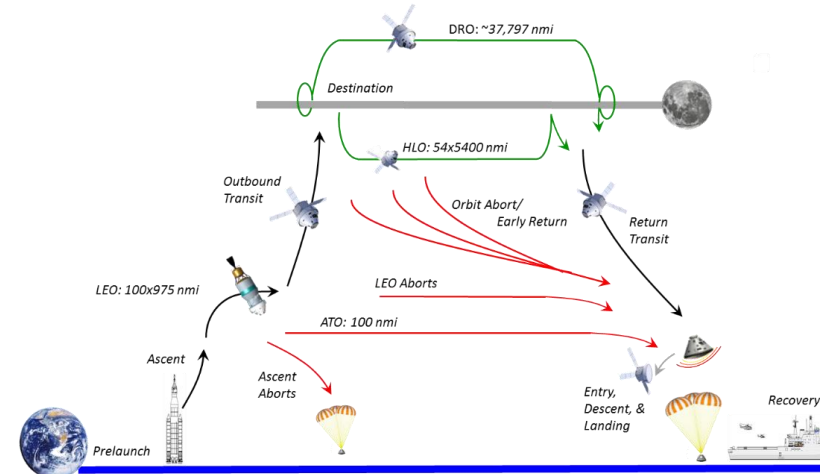
Mission Capabilities



Orion Power System Model

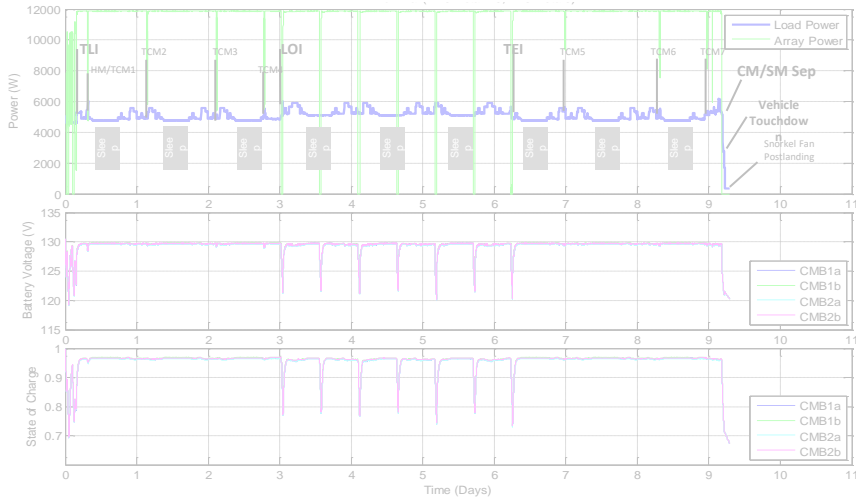


Mission Timelines

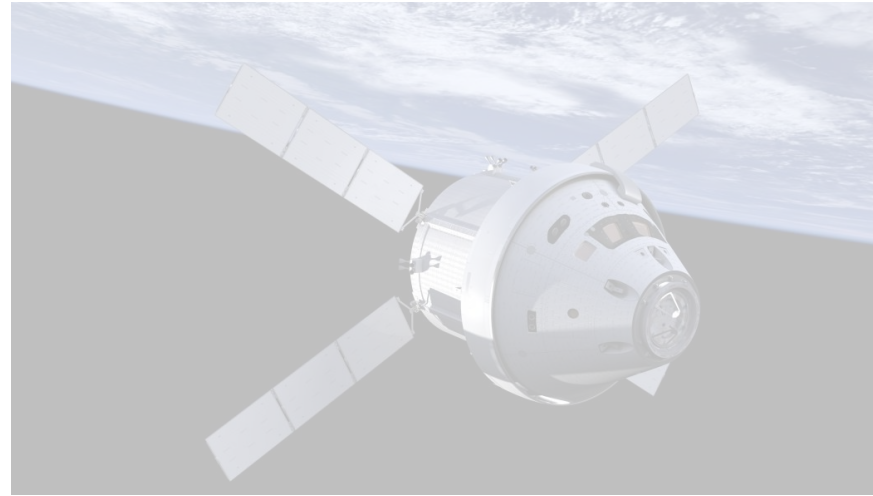


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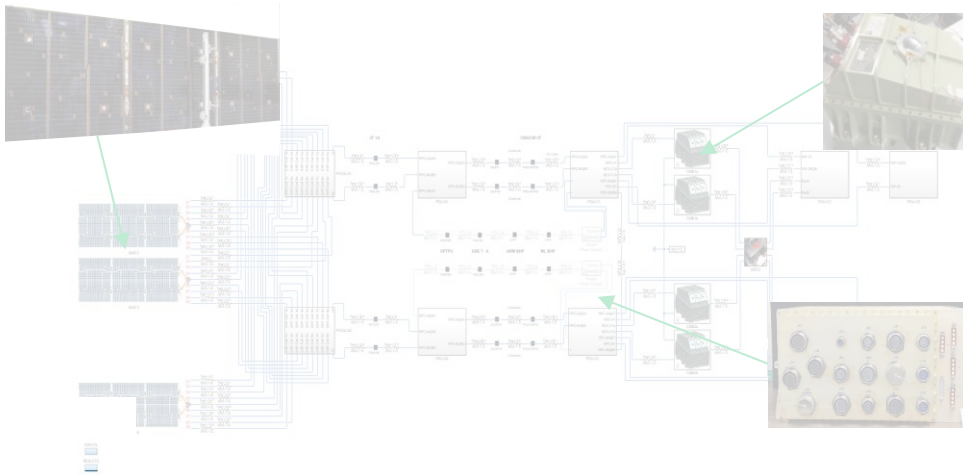
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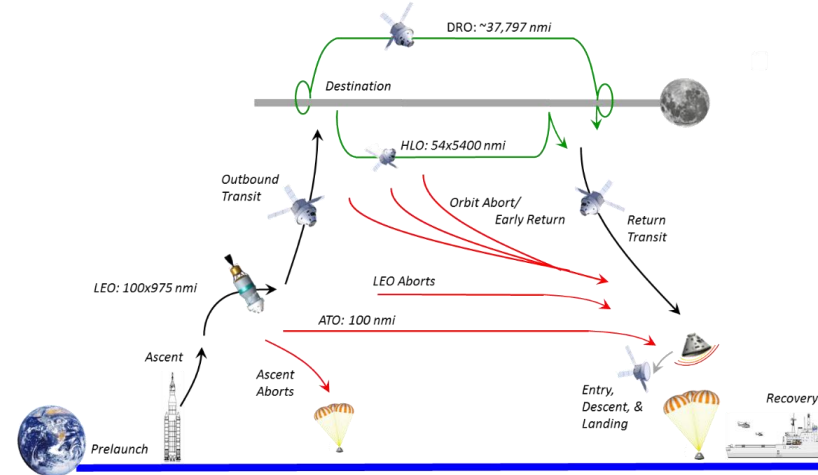
Mission Capabilities



Orion Power System Model

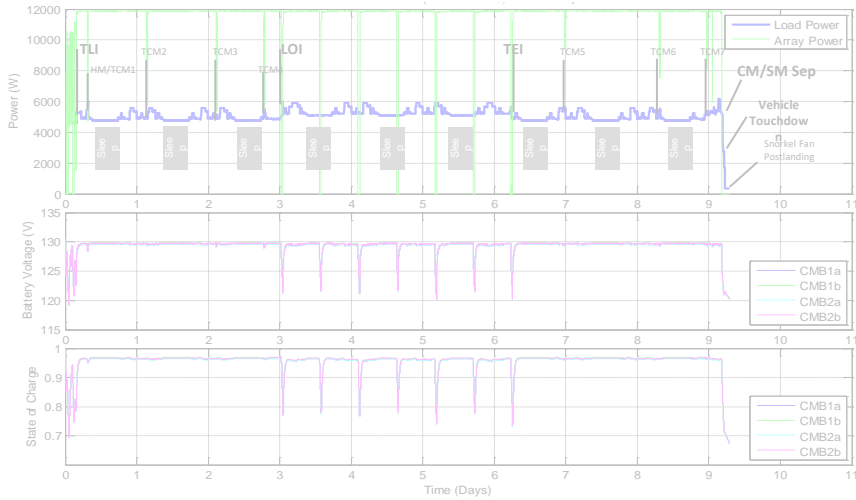


Mission Timelines

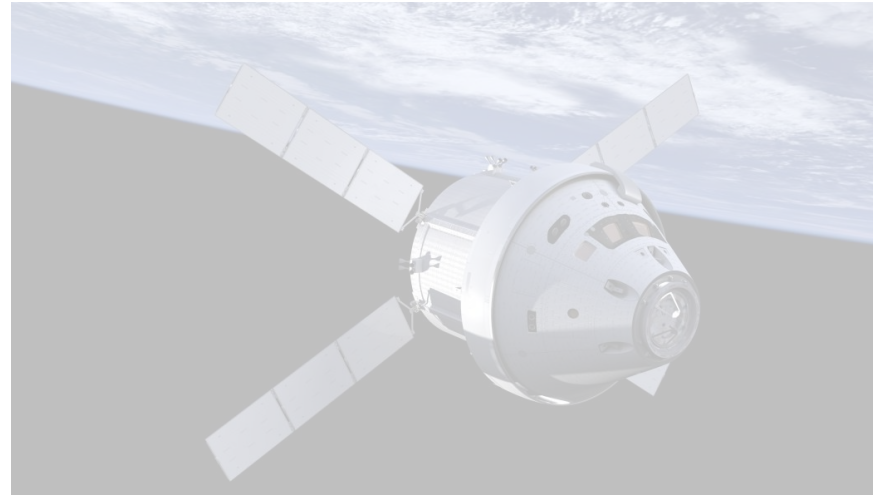


Challenges: Mission Design & System Performance

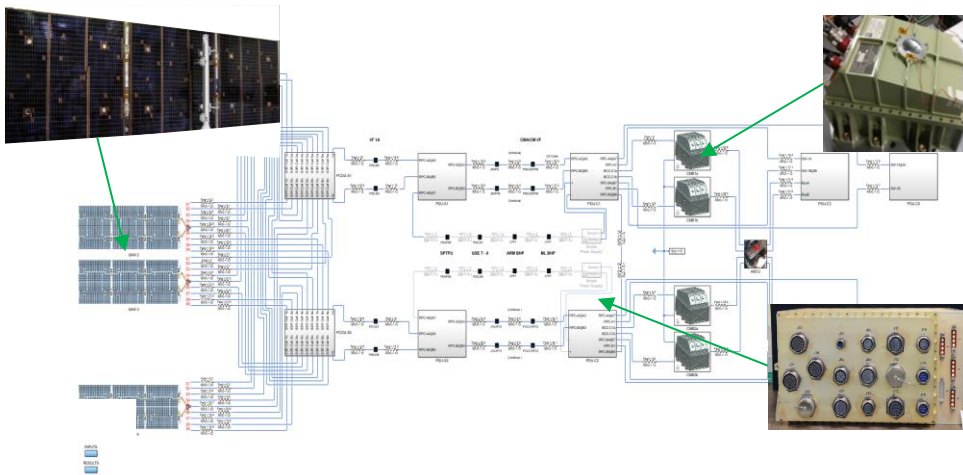
Integrated Power Analysis



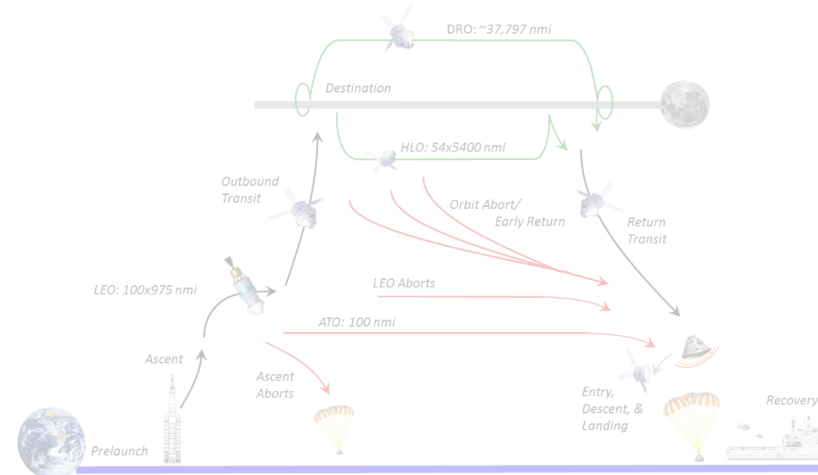
Mission Capabilities



Orion Power System Model

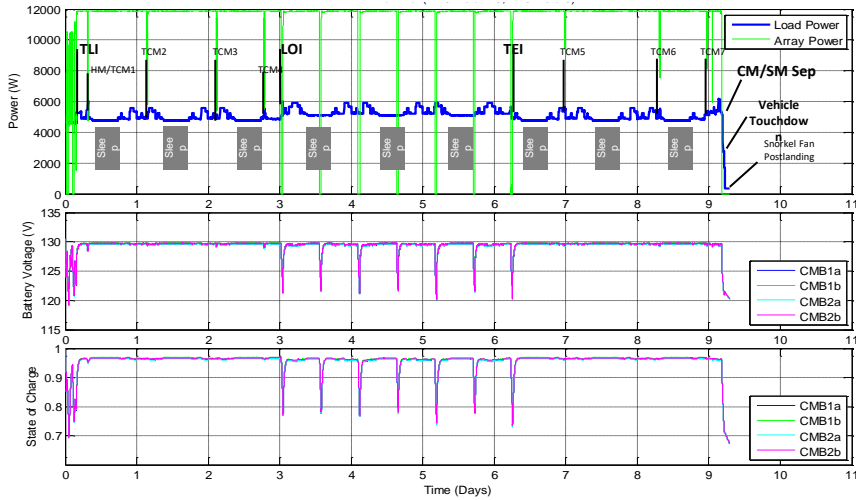


Mission Timelines

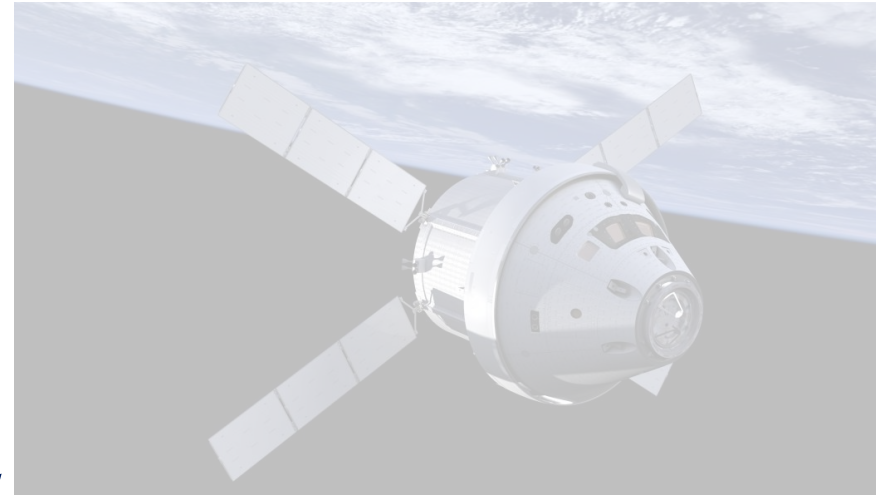


Challenges: Mission Design & System Performance

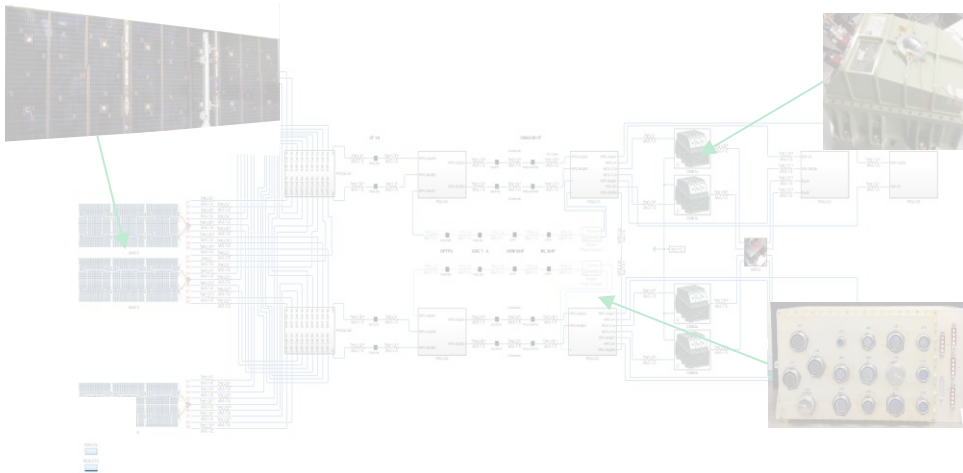
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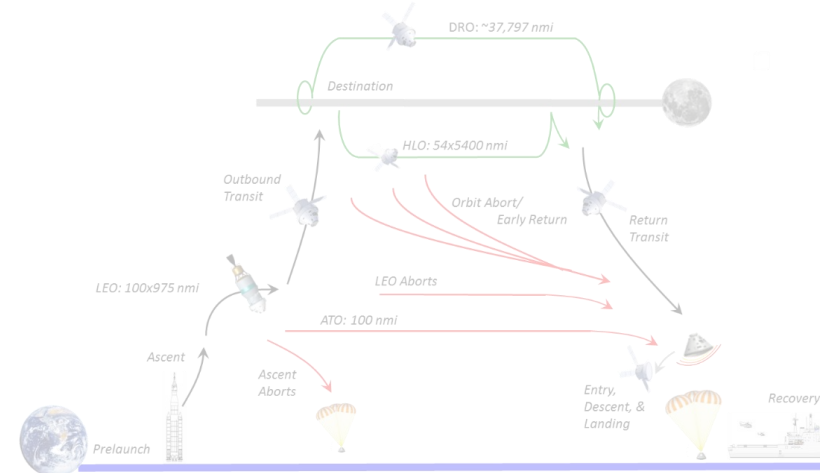
Mission Capabilities



Orion Power System Model

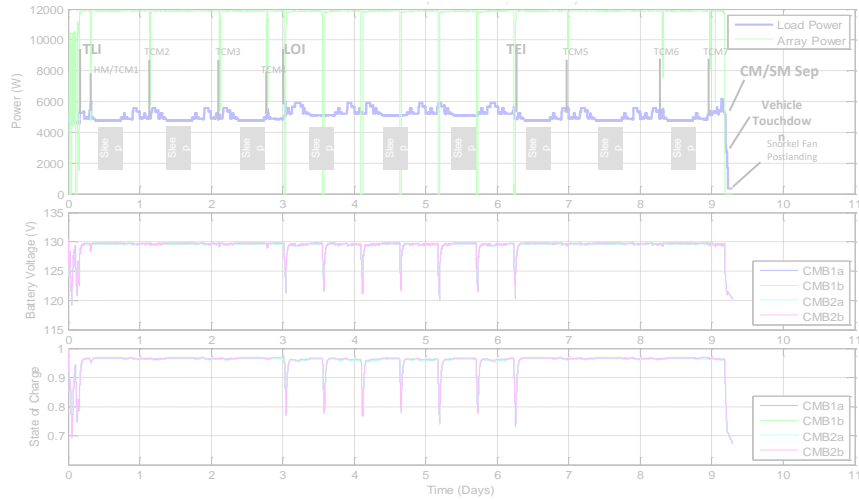


Mission Timelines

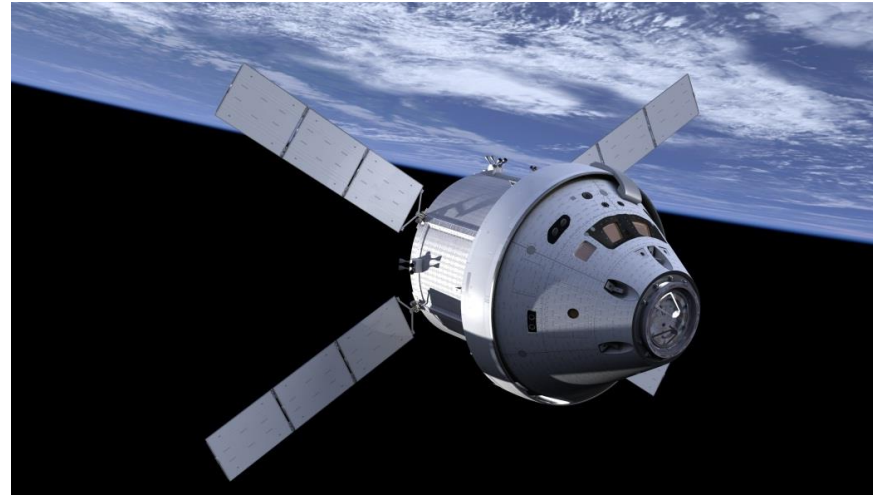


Challenges: Mission Design & System Performance

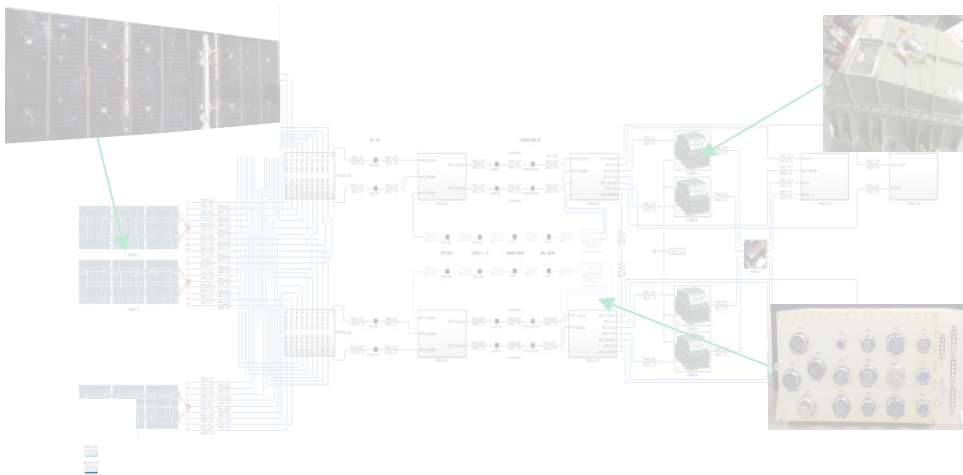
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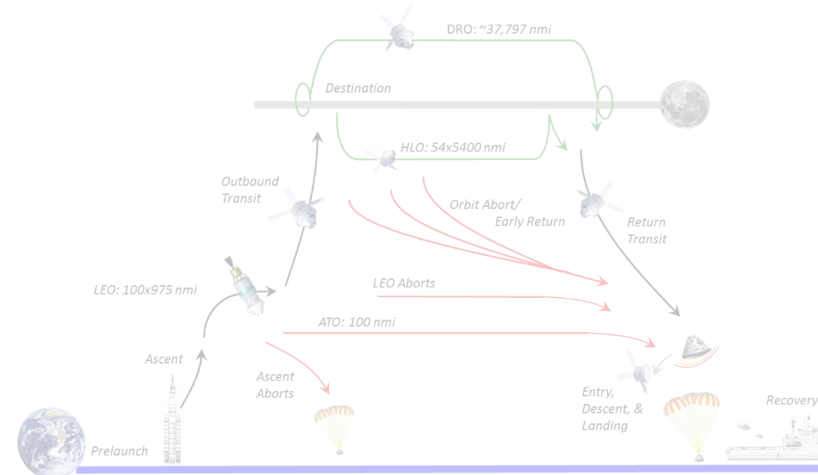
Mission Capabilities



Orion Power System Model

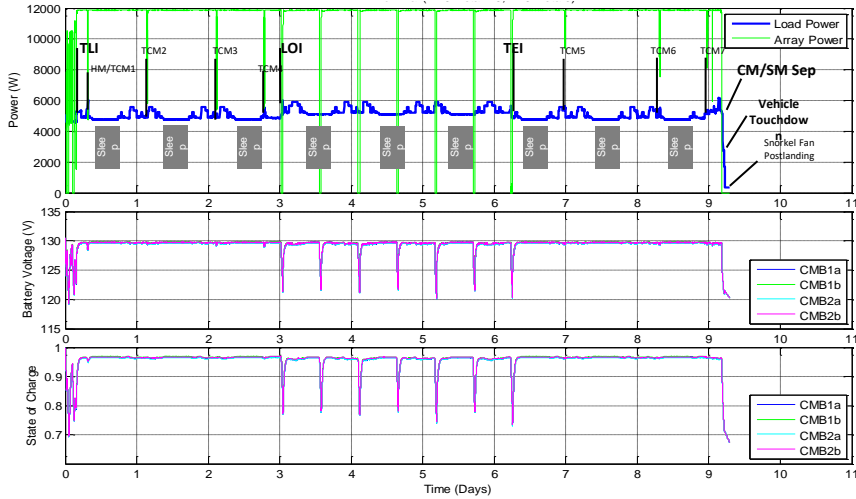


Mission Timelines

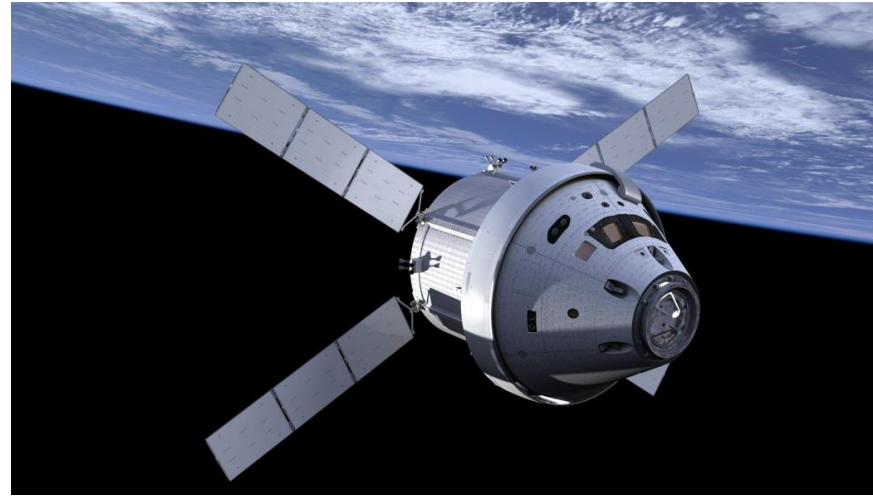


Challenges: Mission Design & System Performance

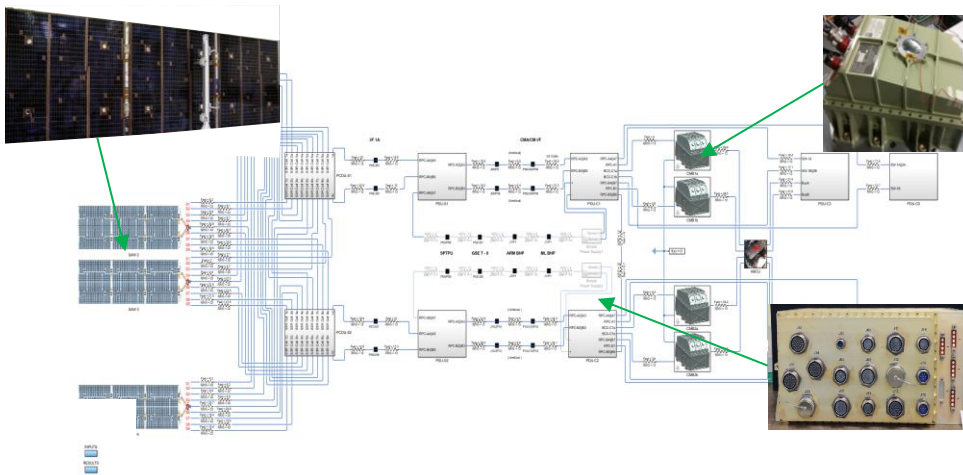
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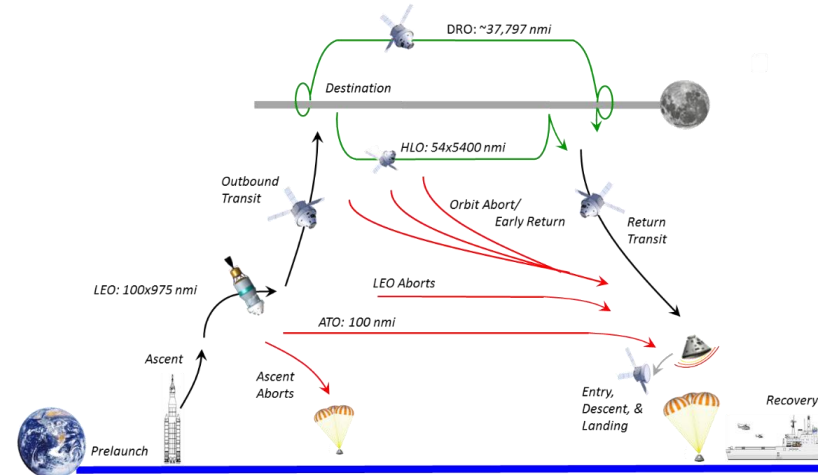
Mission Capabilities



Orion Power System Model



Mission Timelines



Conclusions

- **Accurate modeling of spacecraft power systems is essential to success as our missions increasingly become more complex**
- **SPoC is the integrated power model tool that provides that capability**
 - SPoC continues to be used in support of engineering development and operations of NASA's spaceflight programs and will continue to do so in the future



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Thank you!



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